

The University of Miami Helicopter Observation Platform (HOP)

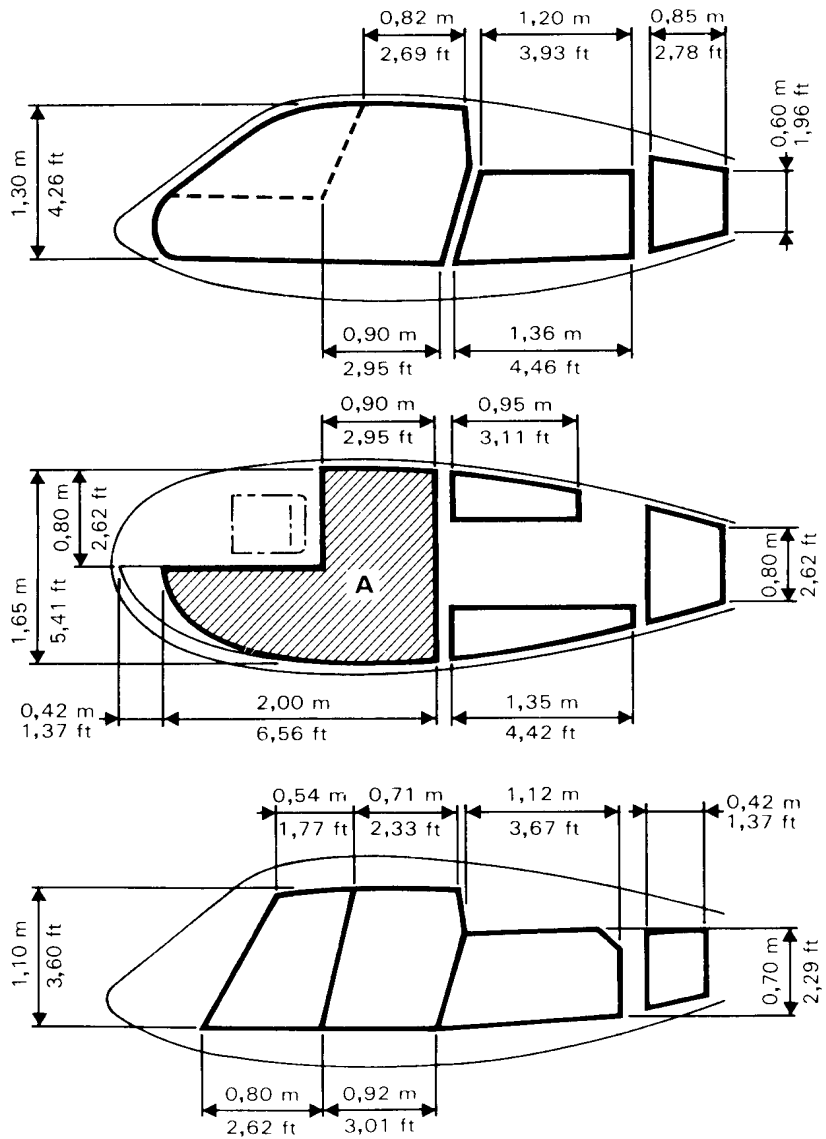
Roni Avissar



Airbus Helicopter H125 - Performance at Max. GROSS WEIGHT, ISA, SL

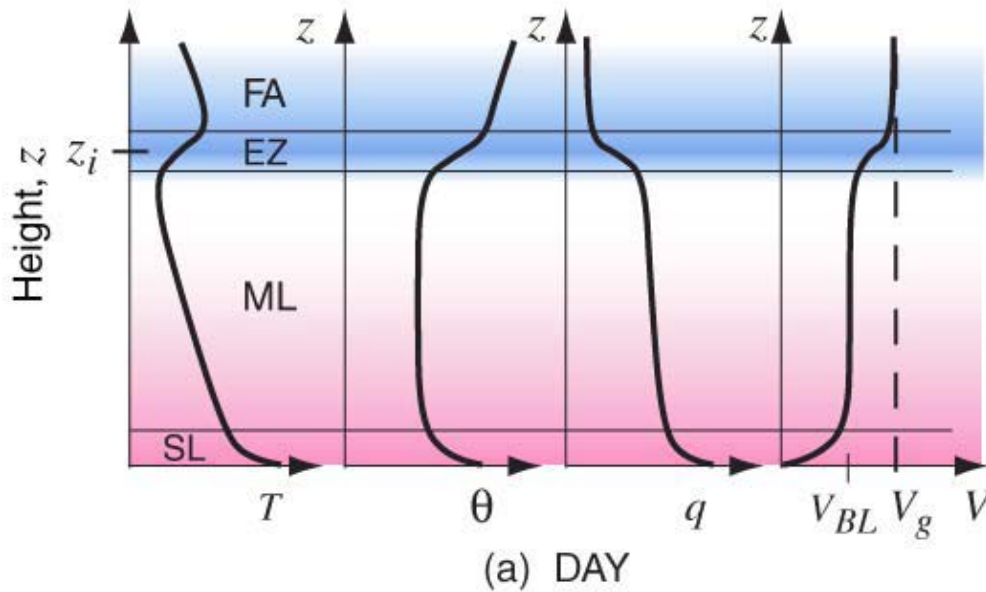
Maximum speed (Vne)	155 kts / 287 km/hr
Fast cruise speed	140 kts / 259 km/hr
Range (@127 kts)	345 nm / 638 km
Max Endurance	4'23" (no reserve) – 4 hrs for science mission
Rate of climb	1,959 ft per min / 10 m/s
Service ceiling	16,550 ft / 5,044 m
Hover ceiling IGE	13,200 ft / 4,023 m
Hover ceiling OGE	11,100 ft / 3,383 m
Maximum Altitude	23,000 ft / 7,010 m (landed on Mt Everest!)
Maximum takeoff weight	5,225 lb / 2,370 kg
Maximum with external load	6,172 lb / 2,800 kg (+950 lb for Payload)
Maximum cargo-swing load	3,086 lb / 1,400 kg
Maximum Scientific Payload	1,000/3000 lb (fully fueled with 2 pilots)



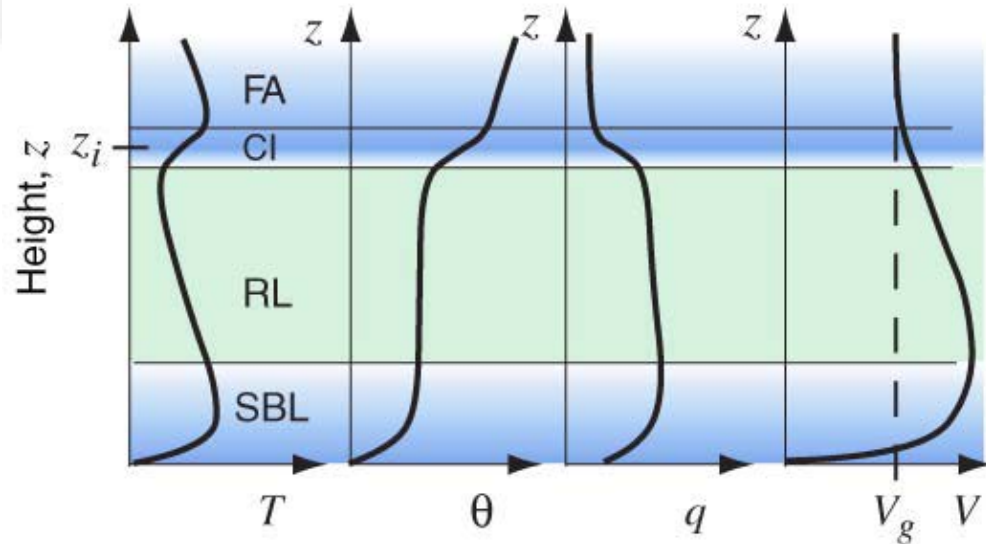


Cabin main dimensions

CABIN	
Surface A	2.60 m ² 27.98 ft ²
Volume	3.000 m ³ 105.94 ft ³
LH HOLD	
Surface	0.43 m ² 4.62 ft ²
Volume	0.235 m ³ 8.29 ft ³
RH HOLD	
Surface	0.35 m ² 3.76 ft ²
Volume	0.200 m ³ 7.06 ft ³
REAR HOLD	
Surface	0.55 m ² 5.92 ft ²
Volume	0.565 m ³ 19.95 ft ³
TOTAL HOLDS	
Surface	1.33 m ² 14.3 ft ²
Volume	1.000 m ³ 35.30 ft ³



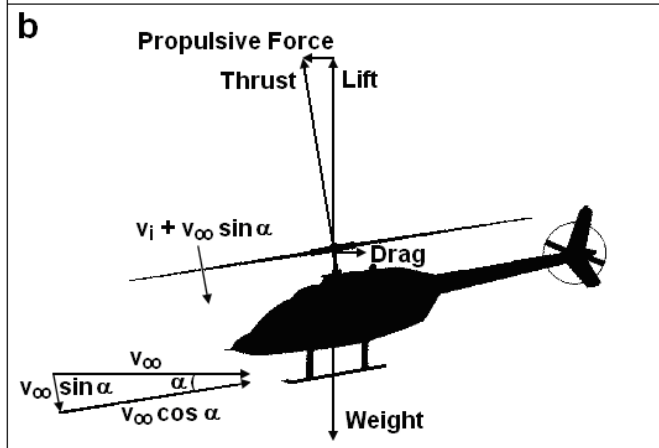
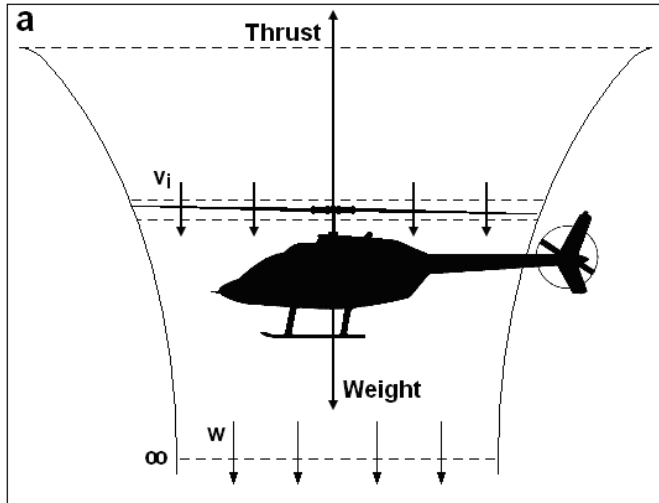
(a) DAY



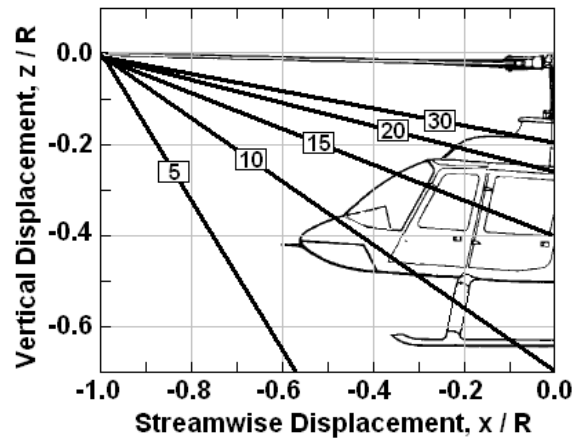
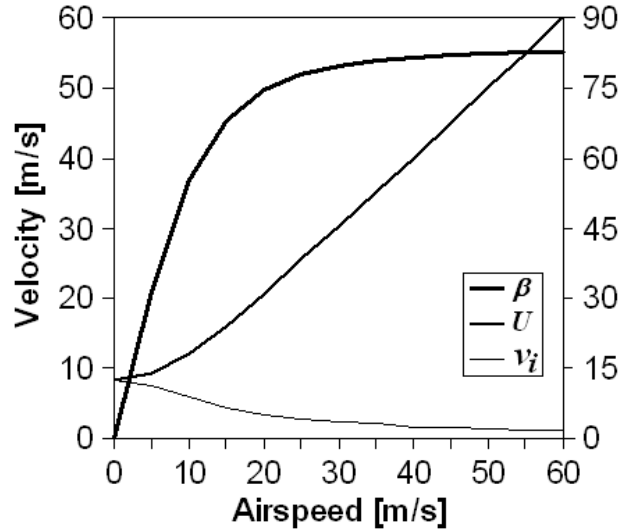
(b) NIGHT

Why a helicopter?

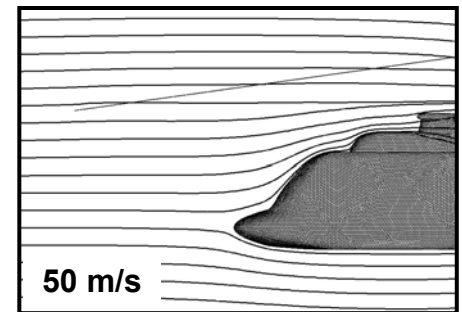
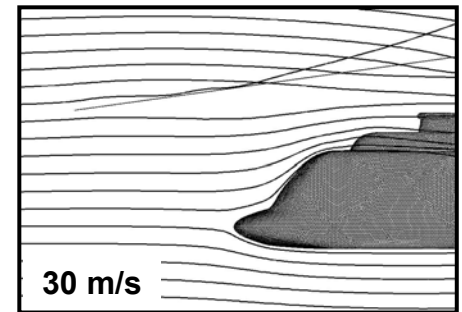
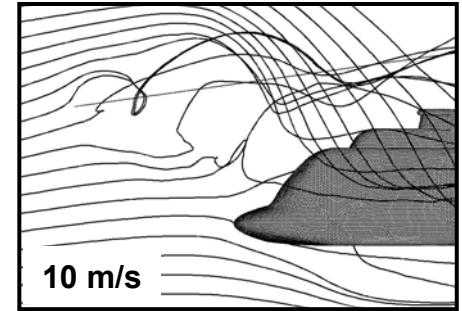
1. Various environmental observations require low altitude, very-high frequency of sampling, and/or slow speed of sampling (e.g., aerosols, which have a key impact on climate and health; fluxes - water, carbon, others)
2. Maneuverability (complex terrain, urban areas, quick turns for flight tracks)
3. Time at station (with a fuel truck on the ground, no need to commute to an airport)
4. Remote location (e.g., operation from a ship)



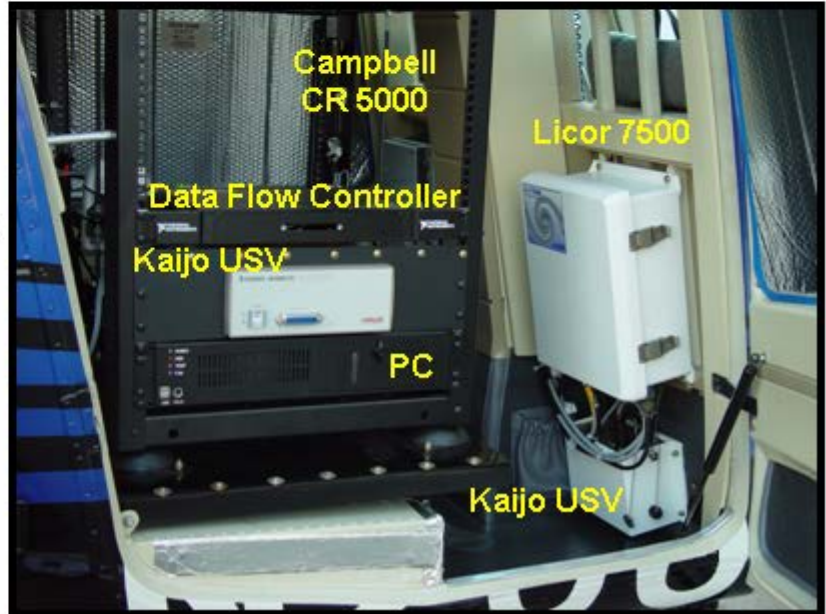
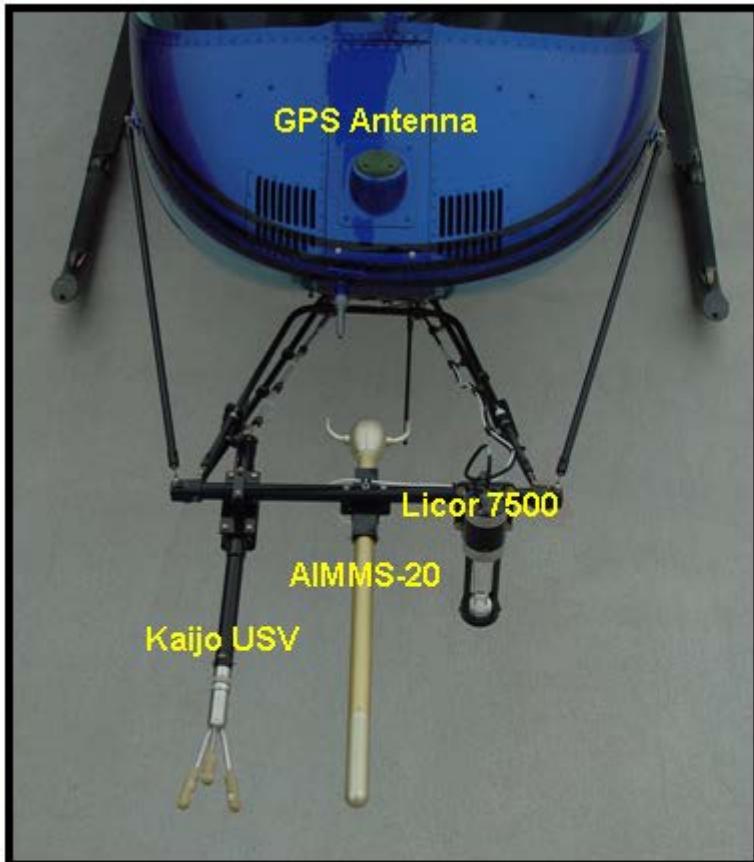
Rotor Wake Skewness

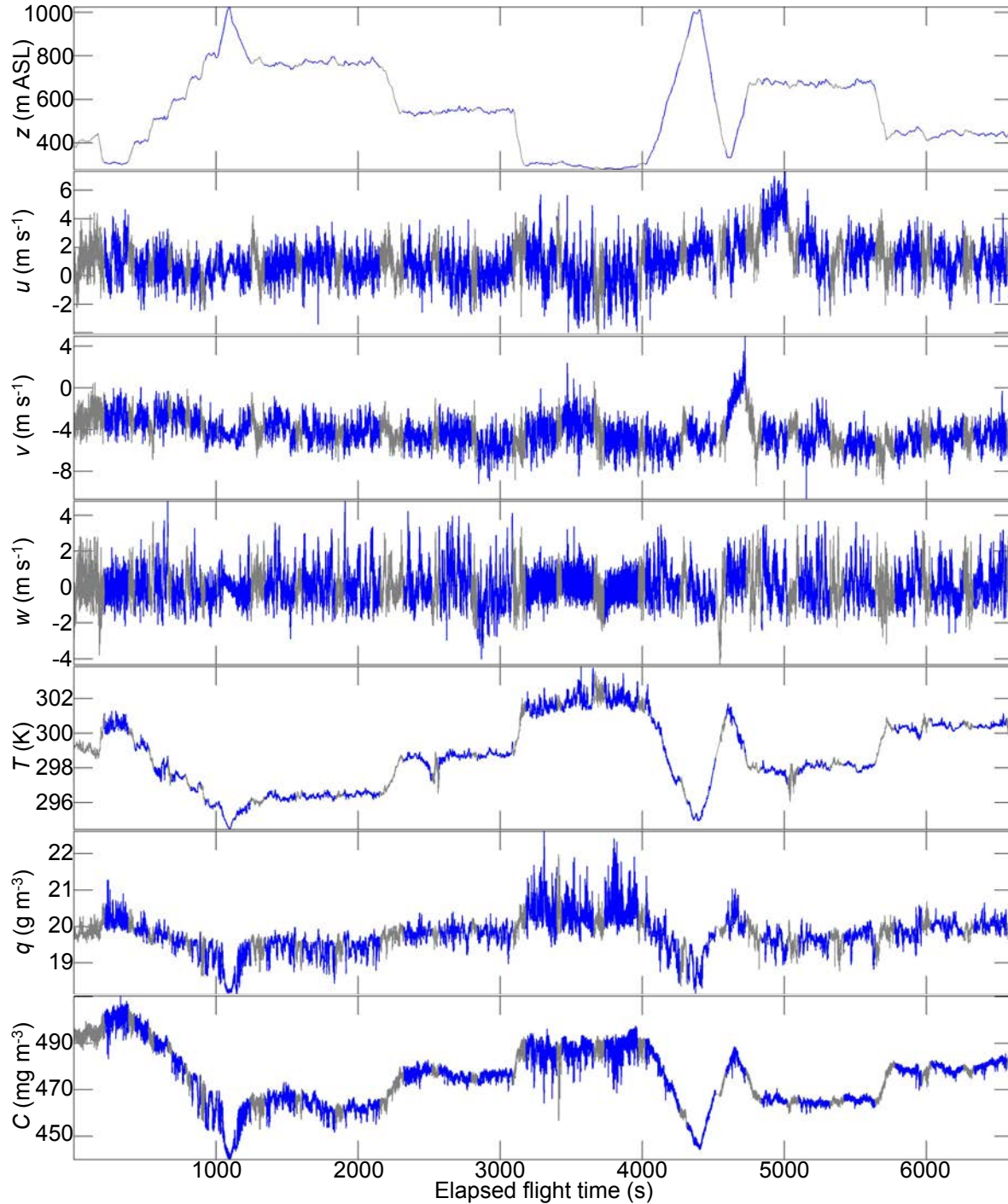


Fluent CFD



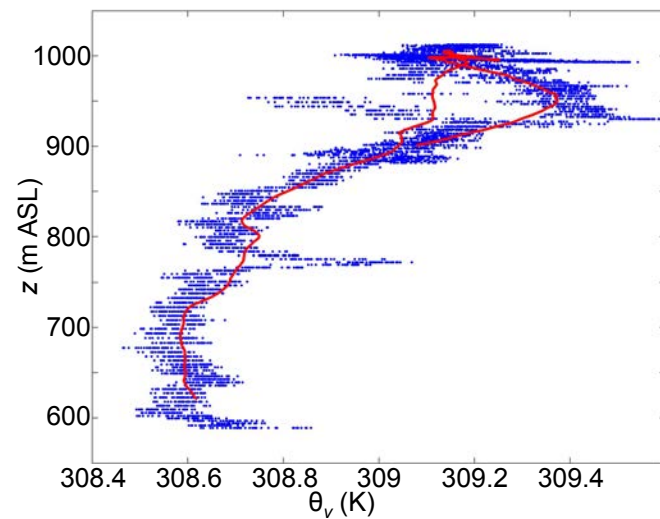
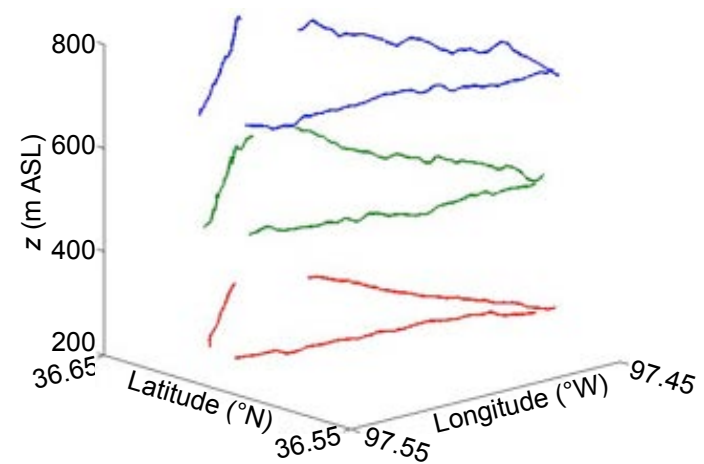
Glauert's (1935) rotor-wake model:
$$U = \sqrt{(v_\infty \cos \alpha)^2 + (v_\infty \sin \alpha + v_i)^2}$$

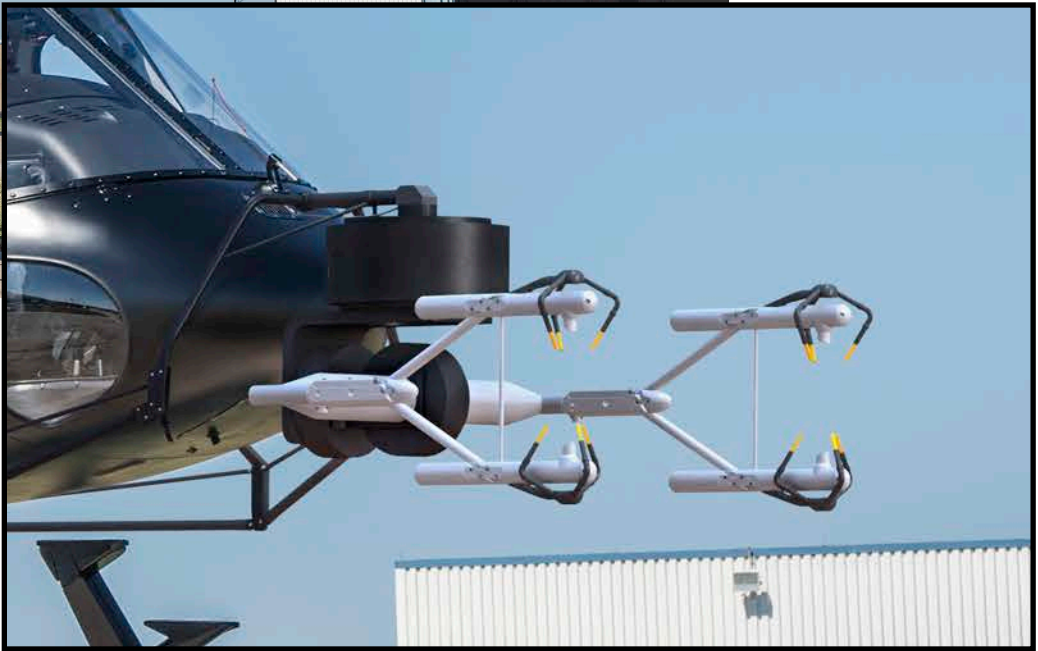


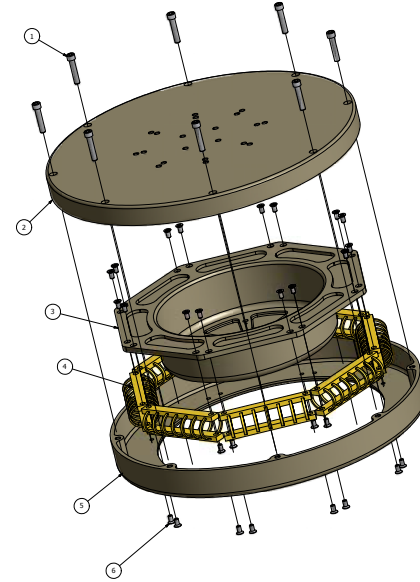
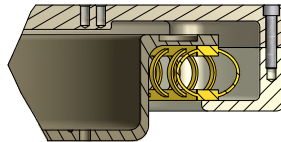
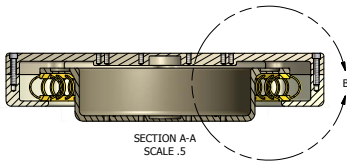
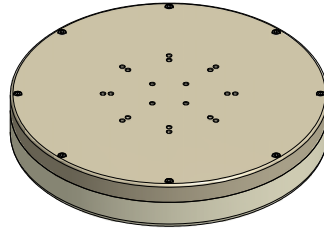
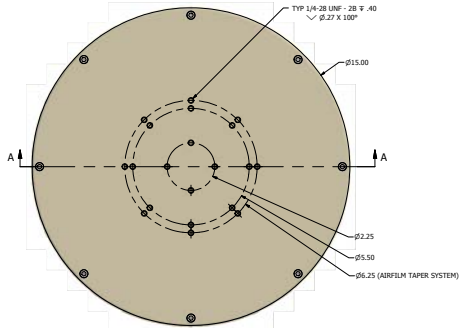


HOP Data

19 June 2007 CLASIC HOP flight







PARTS LIST			
ITEM	QTY	PART NUMBER	DESCRIPTION
1	8	M10-30999-08	CAP SCREW
2	1	10	TOP COVER
3	1	12	STEP RING
4	8	WR3-4000-150	WIRE ROPE ISO
5	1	12	LOWER RING
6	32	M20-6094-555	MACHINE SCREW

ASSY DRAWING
 CHECKED
 SP, Mering 2/6/2002
 APPROVED
 DATE: 2/6/2002
 BY: SP, Mering
 AIR - 10-1000
 AIR - 10-1000
 AIR - 10-1000

AirFilm Camera Systems
 Georgetown, Ca 920-333-0193 www.airfilm.com
15 INCH DIA. VIBRATION
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 DWG NO: **V15-1**
 SHEET 1 OF 2

Conclusions

We demonstrated that HOP is a valuable addition to the research aircraft fleet, especially for land-atmosphere and air-sea interactions. An operational version of the Duke prototype was rebuilt at the University of Miami Rosenstiel School based on the knowledge gained with this prototype. The platform is on schedule to be available for research in Spring 2015.

- Avissar et al, 2009, *Bull. Amer. Met. Soc.*, ., **90**, 939-954.
- Holder, Bolch and Avissar, 2010, *J. Atmos. Ocean Tech.*, **8**, 671-683.



Cost of Operation...

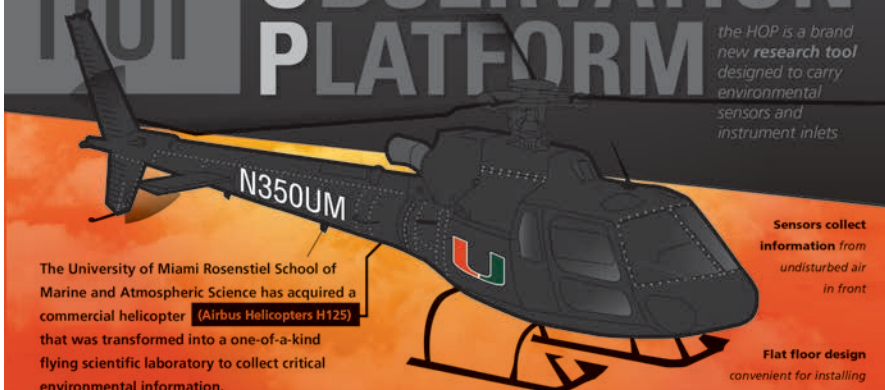
Rate is \$1,995/hr (including ALL costs: insurance, hangar, maintenance and inspections, operation, pilot, fuel, parts and labors). A typical intensive field campaign lasts about 10-12 days. Flying 5 hr/day on average results in 50-60 hours of data collection (plus transportation to the site) and costs about \$150K. Typical field campaign budgets are in millions of \$ and, therefore, the cost of the HOP is typically minor, especially given the uniqueness that it brings to the table...

INTRODUCING THE

HOP

HELICOPTER OBSERVATION PLATFORM

the HOP is a brand new research tool designed to carry environmental sensors and instrument inlets



The University of Miami Rosenstiel School of Marine and Atmospheric Science has acquired a commercial helicopter (Airbus Helicopters H125) that was transformed into a one-of-a-kind flying scientific laboratory to collect critical environmental information.

The HOP fills critical gaps in physical, chemical and biological observations of the environment.

Sensors collect information from undisturbed air in front

Flat floor design convenient for installing scientific payload

HOP OPERATIONAL DETAILS

Scientific payload (Up to)

1,000 LBS
internally

3,000 LBS
externally

Flight time

4 HOURS
without refueling
at airspeed of
65 knots

Fast cruising speed

140 KNOTS

Range

350 NAUTICAL MILES
at top
cruising
speed

Altitude

from a few feet above
the Earth's surface, to
the mid troposphere

works at various altitudes
Ideal to access remote areas and conduct airborne sea-surface measurements.



Capability to hover is

IDEAL FOR REMOTE-SENSING OBSERVATIONS

Capable of

HIGH-ALTITUDE FLIGHTS

Easily transported

ANYWHERE IN THE WORLD

Can be operated from an

OCEANOGRAPHIC RESEARCH SHIP AT SEA

Ideal to

ACCESS REMOTE AREAS AND CONDUCT AIRBORNE SEA-SURFACE MEASUREMENTS

Available to the

ENTIRE SCIENTIFIC COMMUNITY



@UMiamiRSMAS #UMHOP

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